

# **Mathematical Modeling of Spacecraft Gyroscope Noise**

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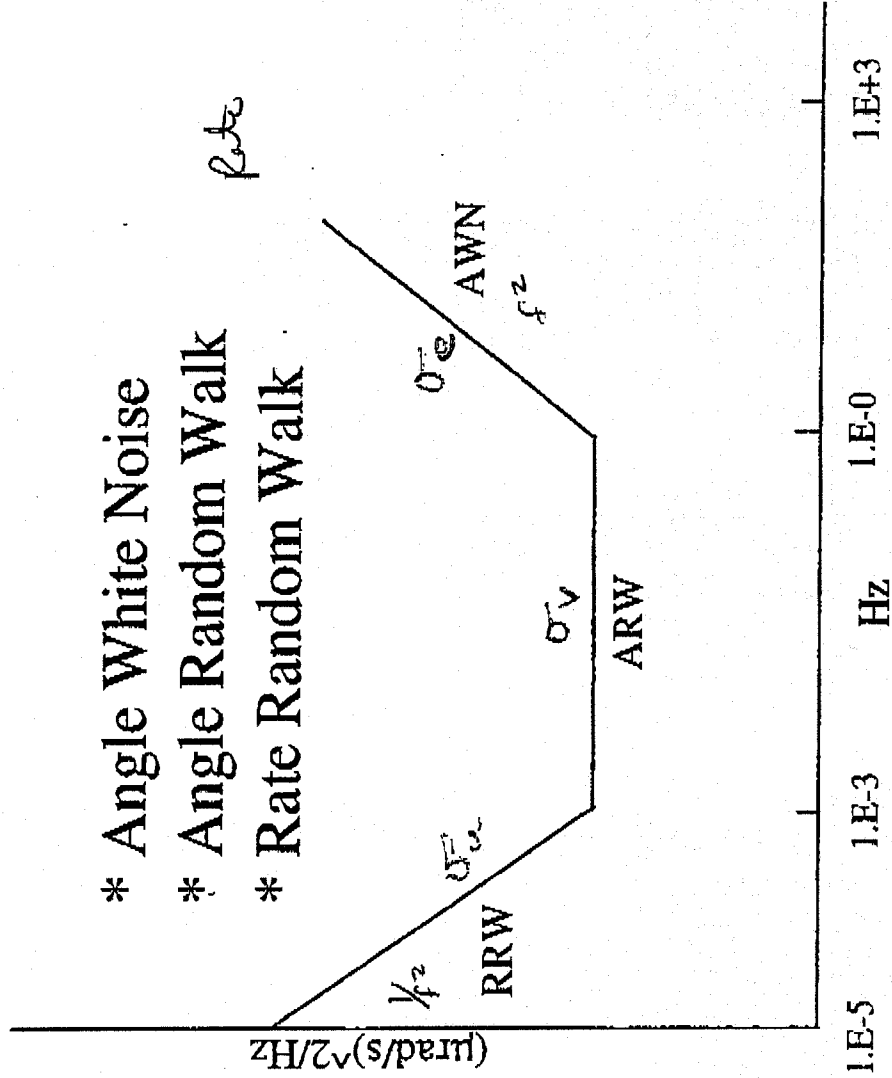
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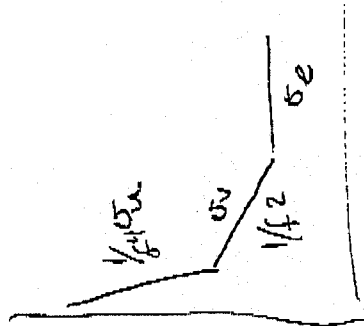
# Gyro Noise Descriptions

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## Power Spectral Density Chart



Angle PSD



# Modeling Gyro Noise

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For high-accuracy applications, most strapdown gyros are rate-integrating gyros (RIGs), that is, they measure (ideally) an integrated angle

$$\Theta = \int_{t_0}^{t_0+\Delta t} \omega_{\text{input}}(t) dt, \text{ where}$$

$\omega_{\text{input}}(t) = \vec{\omega} \cdot \vec{U}_{\text{input}} =$  component of the spacecraft angular velocity along an input axis represented by the unit vector  $\vec{U}_{\text{input}}$ , perpendicular to the gyro spin axis.

In general,  $\vec{U}_{\text{input}} = A_{\text{misalignment}} \vec{U}_{\text{nominal}}$ .

A rate-integrating gyro produces its output by digitally summing the rebalance torques needed to keep the rotor orientation fixed in the spacecraft as the spacecraft rotates.

A rate gyro measures  $\omega_{\text{input}}$  rather than  $\Theta$ .

As we have seen, a gyro is a very complex coupled electromechanical system, and modeling noise sources in gyros from first principles would be very difficult. Fortunately, there is a fairly tractable model that contains a fair approximation to the most important effects. This model was developed by TRW

engineers for the HEAO-A (High Energy Astronomy Observatory spacecraft), most notably Robert L. Fienberg. HEAO-A used single-axis floated gyros but we use the same model for two-axis dry tuned gyros. This ignores errors arising from the coupling of the two axes of the two-axis gyros.

The model is

$$\dot{\Theta}(t) = \omega_{\text{input}}(t) + b(t) + \eta_1(t)$$

where  $b$  is the gyro drift, with time dependence

$$\dot{b}(t) = \eta_2(t)$$

The quantities  $\eta_1(t)$  and  $\eta_2(t)$  are zero-mean white noised processes, which are not quite the same as the random variables we considered before, because they are functions of time. Zero-mean, as before, means that  $E\{\eta_1(t)\} = E\{\eta_2(t)\} = 0$  for all  $t$ , where  $E\{\dots\}$  means expectation value.

The drift  $b(t)$ , whose time derivative is purely white noise, undergoes a random walk. This is mathematically the same as Brownian motion, and is sometimes also called the drunkard's walk.

We need to know about expectation value of product of  $\eta_1$  and  $\eta_2$ . We assume that  $\eta_1$  and  $\eta_2$  are uncorrelated, which means that

$$E\{\eta_1(t)\eta_2(t')\} = 0 \text{ for any } t, t'.$$

Also the "white noise" approximation on  $\eta_1$  and  $\eta_2$  means that the value of each function at a given time is uncorrelated with the value of the same function at a different time.

Thus

$$E\{\eta_1(t)\eta_1(t')\} = \sigma_v^2 \delta(t-t') \text{ and } \sigma_e$$

$$E\{\eta_2(t)\eta_2(t')\} = \sigma_u^2 \delta(t-t')$$

where  $\delta(x)$  is a function that is zero for  $x \neq 0$ . The white noise assumption allows  $\sigma_v^2$  and  $\sigma_u^2$  to be functions of time, but we'll assume for simplicity that they're constant.

We haven't said anything about the value of  $\delta(x)$  at  $x=0$ . We want it to be infinite there, so infinite that

$$\int_a^b \delta(x-y) dx = 1 \quad \text{if } a < y < b$$

Mathematically, such a function doesn't exist, but it can be understood rigorously as a generalized function or a distribution. Physicists call  $\delta(x)$  the Dirac delta function and engineers call it the unit impulse function. For real gyros we have, instead of  $\delta(x)$ , a function that has a narrow high peak at  $x=0$ ; this is colored noise. White noise is easier to analyze.

For computations below we need to know  $\int_0^a \int_0^b \delta(x-y) dx dy$

If  $b > a$  we integrate over  $x$  first using  $\int_0^b \delta(x-y) dx = 1$ , since  $0 < y < a < b$

If  $a > b$  we integrate over  $y$  first, using  $\int_0^a \delta(x-y) dy = 1$ , since  $0 < x < b < a$

We are left with the integral of 1 over a range from 0 to the minimum of  $a$  and  $b$ . Thus  $\int_0^a \int_0^b \delta(x-y) dx dy = \text{minimum of } (a, b)$ .

Now back to the gyros. From the  $b$  equation

$$b = b_0 + \int_0^{\Delta t} \eta_2(t) dt \quad \text{where we've taken } t_0 = 0 \text{ as the lower limit of integration to simplify the notation.}$$

Then  $E\{b\} = b_0$  since  $E\{\eta_2\} = 0$ .

$$\Rightarrow b^2 = b_0^2 + 2b_0 \int_0^{\Delta t} \eta_2(t) dt + \int_0^{\Delta t} \eta_2(t) dt \int_0^{\Delta t} \eta_2(t') dt'$$

$$E\{b^2\} = b_0^2 + E\left\{\int_0^{\Delta t} \eta_2(t) dt \int_0^{\Delta t} \eta_2(t') dt'\right\}$$

$$= b_0^2 + \sigma_u^2 \int_0^{\Delta t} \int_0^{\Delta t} \delta(t-t') dt dt' = b_0^2 + \sigma_u^2 \Delta t$$

To model the drift in a simulation, we can take

$$b = b_0 + \sigma_u (\Delta t)^{1/2} \mathcal{E}_b$$

where  $\mathcal{E}_b$  is a gaussian-distributed random variable with mean 0 and standard deviation unity. It can be seen that this gives the correct expectation values  $E\{b\}$  and  $E\{b^2\}$ .

The gyro output  $\Theta$  is given by

$$\begin{aligned}\Theta &= \Theta_0 + \int_0^{\Delta t} (\omega_{\text{input}} + b + \eta_1) dt = \Theta_0 + \int_0^{\Delta t} \omega_{\text{input}} dt + \int_0^{\Delta t} \left[ b_0 + \int_0^t \eta_2(\tau) d\tau \right] dt \\ &\quad + \int_0^{\Delta t} \eta_1(t) dt \\ &= \Theta_{\text{perfect}} + b_0 \Delta t + \int_0^{\Delta t} \int_0^t \eta_2(\tau) d\tau dt + \int_0^{\Delta t} \eta_1(t) dt\end{aligned}$$

where  $\Theta_{\text{perfect}} \equiv \Theta_0 + \int_0^{\Delta t} \omega_{\text{input}} dt$  is the value of the gyro output in the absence of drifts and noise.

$$E\{\Theta\} = \Theta_{\text{perfect}} + b_0 \Delta t$$

$$E\{b\Theta\} = E\left\{ \left[ b_0 + \int_0^{\Delta t} \eta_2(t) dt \right] \left[ \Theta_{\text{perfect}} + b_0 \Delta t + \int_0^{\Delta t} \int_0^t \eta_2(\tau) d\tau dt + \int_0^{\Delta t} \eta_1(t) dt \right] \right\}$$

$$= b_0 (\Theta_{\text{perfect}} + b_0 \Delta t) + E\left\{ \int_0^{\Delta t} \eta_2(t') dt' \int_0^{\Delta t} \int_0^t \eta_2(\tau) d\tau dt \right\}$$

$$= b_0 (\Theta_{\text{perfect}} + b_0 \Delta t) + \sigma_u^2 \int_0^{\Delta t} \int_0^{\Delta t} \int_0^t \delta(t' - \tau) d\tau dt' dt$$

$$= b_0 (\Theta_{\text{perfect}} + b_0 \Delta t) + \sigma_u^2 \int_0^{\Delta t} t dt = b_0 (\Theta_{\text{perfect}} + b_0 \Delta t) + \sigma_u^2 (\Delta t)^2 / 2$$

Since  $E\{b_b\} = 0$  and  $E\{b_b^2\} = 1$ ,

these equations are satisfied if we model  $\Theta$  in a simulation as

$$\Theta = \Theta_{\text{perfect}} + b_0 \Delta t + (1/2) \sigma_u \Delta t^{3/2} \xi_b + Z = \Theta_{\text{perfect}} + (b_0 + b) \Delta t / 2 + Z$$

where  $Z$  is a random variable with zero mean that is uncorrelated with  $\xi_b$

If we get  $E\{\Theta^2\}$  right, we've got all the expectation values of things linear and quadratic in  $\Theta$  and  $b$ .



$$E\{\theta^2\} = (\theta_{\text{perfect}} + b_0 \Delta t)^2 + E\left\{\int_0^{\Delta t} \int_0^t \eta_2(\tau) d\tau dt \int_0^{\Delta t} \int_0^{t'} \eta_2(\tau') d\tau' dt'\right\} \\ + E\left\{\int_0^{\Delta t} \eta_1(t) dt \int_0^{\Delta t} \eta_1(t') dt'\right\}$$

Since  $E\{\eta_1\} = E\{\eta_2\} = E\{\eta_1 \eta_2\} = 0$

$$E\{\theta^2\} = (\theta_{\text{perfect}} + b_0 \Delta t)^2 + \sigma_u^2 \int_0^{\Delta t} \int_0^{\Delta t} \int_0^t \int_0^{t'} \delta(\tau - \tau') d\tau' d\tau dt' dt \\ + \sigma_v^2 \int_0^{\Delta t} \int_0^{\Delta t} \delta(t - t') dt dt' \\ = (\theta_{\text{perfect}} + b_0 \Delta t)^2 + \sigma_u^2 \int_0^{\Delta t} \int_0^{\Delta t} [\text{minimum of } (t, t')] dt' dt + \sigma_v^2 \Delta t$$

The remaining integral can be split into parts with  $t' < t$  and  $t' > t$ .

$$\int_0^{\Delta t} \left[ \int_0^t t' dt' + \int_t^{\Delta t} -t dt' \right] dt = \int_0^{\Delta t} \left[ t^2/2 + t(\Delta t - t) \right] dt$$

$$= \Delta t \int_0^{\Delta t} t dt - (1/2) \int_0^{\Delta t} t^2 dt = \Delta t^3 \left[ (1/2) - (1/6) \right] = \Delta t^3 / 3$$

Thus  $E\{\theta^2\} = (\theta_{\text{perfect}} + b_0 \Delta t)^2 + \sigma_u^2 (\Delta t)^3 / 3 + \sigma_v^2 \Delta t$ .

But our equation for modeling  $\theta$  in a simulation gives

$$E\{\theta^2\} = (\theta_{\text{perfect}} + b_0 \Delta t)^2 + \sigma_u^2 (\Delta t)^3 / 4 + E\{Z^2\}$$

Comparing these shows that we must have

$$E\{Z^2\} = \sigma_v^2 \Delta t + \sigma_u^2 (\Delta t)^3 / 12$$

This can be accomplished by putting

$$Z = \sqrt{\sigma_v^2 \Delta t + \sigma_u^2 (\Delta t)^3 / 12} \xi_0$$

where  $\xi_0$  is a gaussian-distributed random variable with zero mean and standard deviation unity, uncorrelated with  $\xi_b$ .

Thus we need two calls to a Gaussian random-number generator to get  $\xi_b$  and  $\xi_0$ , after which

$$b = b_0 + \sigma_u(\Delta t)^{1/2} \xi_b \rightarrow \theta = \theta_{\text{perf}} + b_0 \Delta t + \sqrt{\frac{\sigma_u^2 \Delta t^3}{4}} + \sqrt{\sigma_v^2 \Delta t + \sigma_u^2 (\Delta t)^3 / 12} \xi_0$$

$$\theta = \theta_{\text{perfect}} + (b_0 + b) \Delta t / 2 + \sqrt{\sigma_v^2 \Delta t + \sigma_u^2 (\Delta t)^3 / 12} \xi_0$$

We encode  $\theta$  for output, including scale factor error, output noise, and quantization, if we wish. Then  $b$  and  $\theta$  become  $b_0$  and  $\theta_0$  for the next step. If the standard deviation of the output noise is denoted by  $\sigma_e$ , we recover Fallon's result, equation (7-143) in Wertz.

If the rates are slowly varying we can use the trapezoid approximation to integrate  $\theta_{\text{perfect}}$

$$\theta_{\text{perfect}} = \theta_0 + \int_0^{\Delta t} \omega_{\text{input}} dt \approx \theta_0 + (\omega_{\text{input},0} + \omega_{\text{input}}) \Delta t / 2, \text{ giving}$$

$$\theta = \theta_0 + [(\omega_{\text{input}} + b)_0 + (\omega_{\text{input}} + b)] \Delta t / 2 + \sqrt{\sigma_v^2 \Delta t + \sigma_u^2 \Delta t^3 / 12} \xi_0$$

For some rate-integrating gyro packages (DR12V II, for example) the electronics computes and outputs an analog rate

$$\omega_{\text{gyro}} = \frac{\theta - \theta_0}{\Delta t} = [(\omega_{\text{input}} + b)_0 + (\omega_{\text{input}} + b)] / 2 + \sqrt{\sigma_v^2 / \Delta t + \sigma_u^2 \Delta t / 12} \xi_0$$

Most precise ACS use the integrated output  $\theta$  rather than  $\omega_{\text{gyro}}$ , however. The analog rate signal mode is needed by analog "safe hold" control systems.

The expression for  $W_{\text{gyro}}$  is correct if the  $\Delta t$  at which the gyro electronics updates the rate output is the same as the  $\Delta t$  of the simulation. It could be (and often is) that the cycle time of the gyro electronics, which we shall call  $\tau$ , is much less than the OBC  $\Delta t$ . In this case

$$W_{\text{gyro}} = [(w_{\text{input}} + b)' + (w_{\text{input}} + b)]/2 + \sqrt{\sigma_v^2/\tau + \sigma_u^2\tau/12} \mathcal{E}_x$$

where the prime indicates time  $t - \tau$ , and  $\mathcal{E}_x$  is some other zero mean gaussian random variable with standard deviation 1.

We don't know the quantities at  $t - \tau$ , so we use

$$w_{\text{input}}' \approx w_{\text{input}}$$

$$b' = b + \sigma_u \tau^{1/2} \mathcal{E}_y$$

Then

$$W_{\text{gyro}} = w_{\text{input}} + b + \left( \frac{1}{2} \sigma_u \tau^{1/2} \mathcal{E}_y + \sqrt{\sigma_v^2/\tau + \sigma_u^2\tau/12} \mathcal{E}_x \right)$$

The quantity in parentheses is zero mean random noise with variance (assuming  $\mathcal{E}_x$  and  $\mathcal{E}_y$  are uncorrelated)

$$\text{variance} = \frac{\sigma_u^2 \tau}{4} + \frac{\sigma_v^2}{\tau} + \frac{\sigma_u^2 \tau}{12} = \frac{\sigma_v^2}{\tau} + \frac{\sigma_u^2 \tau}{3}$$

Thus we can write

$$W_{\text{gyro}} = w_{\text{input}} + b + \sqrt{\sigma_v^2/\tau + \sigma_u^2\tau/3} \mathcal{E}_w$$

In this case, we need three random number calls, for  $\mathcal{E}_b$ ,  $\mathcal{E}_\sigma$ , and  $\mathcal{E}_w$ .

Minimizing the noise term wrt  $\tau$  gives  $\tau = \sqrt{3} \sigma_v / \sigma_u$  and

$$W_{\text{gyro}} = w_{\text{input}} + b + \sqrt{2\sigma_u\sigma_v/\sqrt{3}} \mathcal{E}_w$$

F. Fairbank's example (JGC V.1.1, no. 4)

$$\sigma_n = 4.81 \times 10^{-5} \text{ sec/s}^{1/2}, \quad \sigma_r = 0.200 \text{ sec/s}^{1/2}$$

$$\tau_{\text{optimal}} = \frac{\sqrt{2} \times 2}{4.81 \times 10^{-5}} = 7202 \text{ sec}$$

In practice  $\tau$  is much less than this and  $\sigma_r$  must be the dominant error term. Say  $\tau = 32 \text{ msec}$ , then

$$\sigma_r^2/\tau + \sigma_n^2\tau/3 = (1.25 + 2.5 \times 10^{-11}) (\text{sec/s})^2$$

$$\begin{aligned} \text{So } \sqrt{\sigma_r^2/\tau + \sigma_n^2\tau/3} &\approx \sqrt{1.25} \text{ sec/s} = 1.12 \text{ sec/s} & (\tau = 32 \text{ ms}) \\ &= 0.56 \text{ " } & (\tau = 128 \text{ ms}) \\ &= 0.28 \text{ " } & (\tau = 512 \text{ ms}) \end{aligned}$$

1 sec/s = 1 deg/hour, which is OK for control